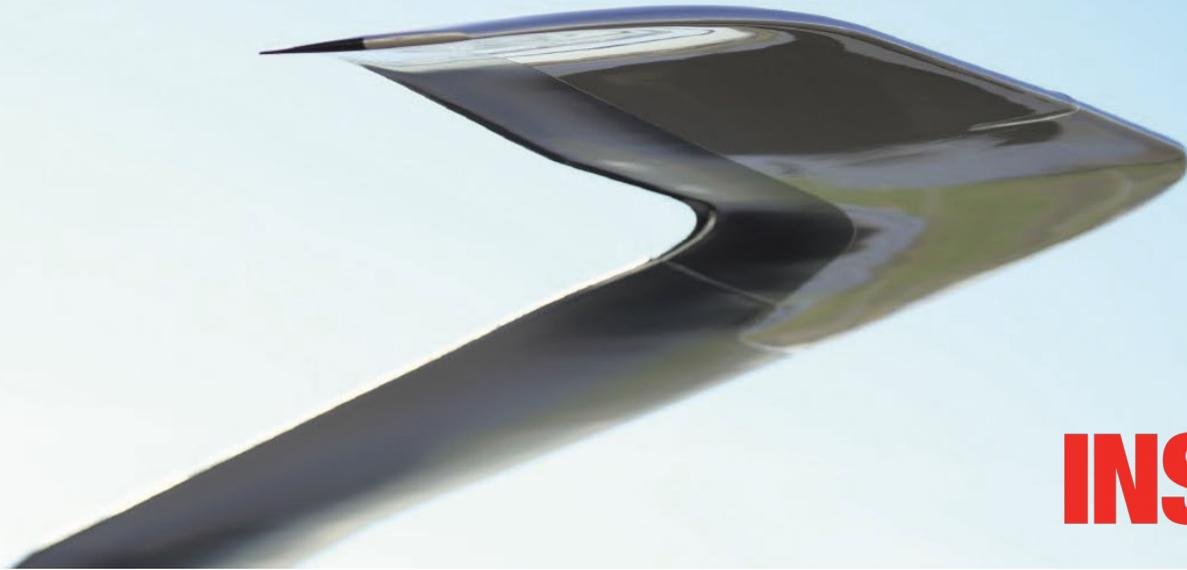




Club Cast3M 2020

November 27, 2020



HELICOPTERS

## A SAM / X-FEM Coupling Approach for Numerical Simulation of Three-Dimensional Crack Propagation under Rolling Contact Fatigue

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**AIRBUS**

# Context & Motivations

- Work focus on the Rolling Contact Fatigue (RCF) crack growth in mechanical components of helicopter MGB,
- Due to the cyclic passages of rolling elements, cracks can initiate on bearing raceway and then propagate,
- Propagation scenarios for the main crack:
  - Propagation towards the surface causing spalling phenomenon,
  - Deep subsurface propagation leading to complete bearing failure.

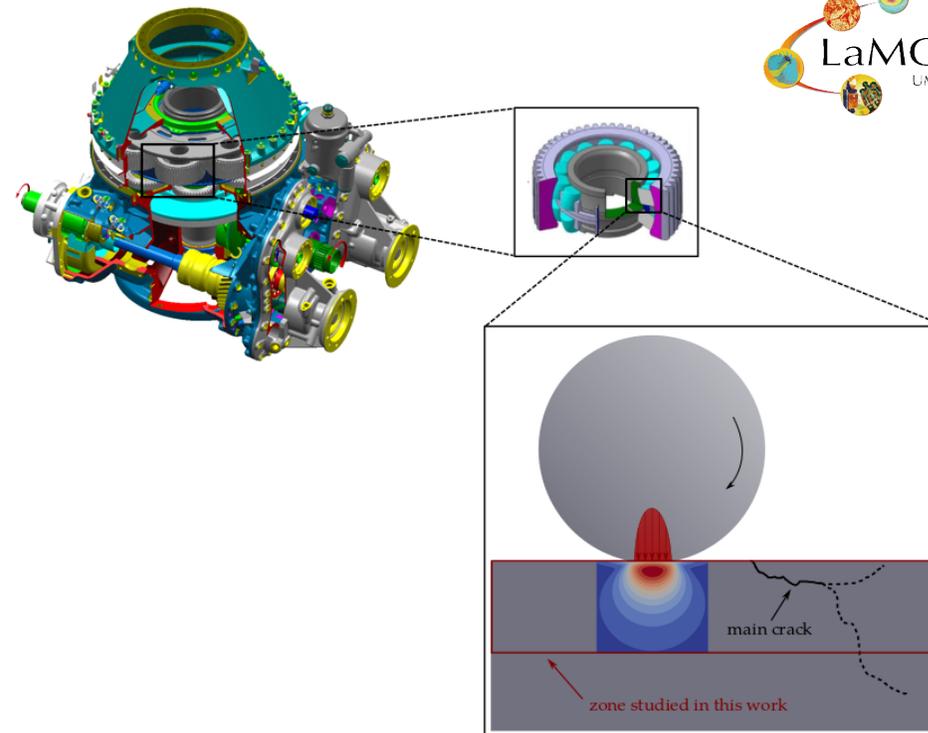


Illustration of rolling contact fatigue crack problem in planet gear

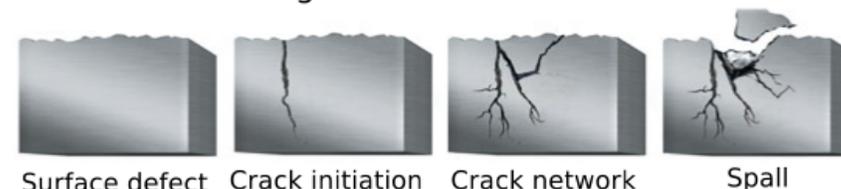


RCF of inner raceway of spherical roller bearing [OLV 05]



Spall initiated at surface [MAR 67]

## RCF damage mechanism from surface



Surface initiated damage from RCF [ZAR 12]

[MAR 67] Martin J. A., Eberhardt A. D. *Identification of potential failure nuclei in rolling contact fatigue*. Journal of Basic Engineering, vol. 89 (4), 1967, p. 932-942.

[OLV 05] Olver AV. *The Mechanism of Rolling Contact Fatigue: An Update*. Proceedings of the Institution of Mechanical Engineers, Part J: Journal of Engineering Tribology, vol. 219 (5), 2005, p. 313-330.

[ZAR 12] Zaretsky E. *How to Determine Bearing System Life*. Rapport, 2012, Machinery Lubrication.

# Context & Motivations

## • Main objectives:

- Modeling the complex 3D crack growth behavior induced by contact fatigue,
- Predict and prevent the ruin of mechanical components such as bearings.



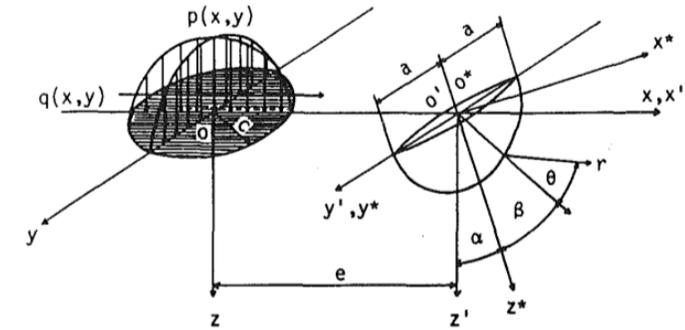
This work is focused only on the propagation stage, the crack initiation is not investigated here.

## • Actual limitations:

- Most of the studies are carried out with 2D models,
- 3D simulations limited to simple configurations or/and very laborious + time consuming.

## • Proposed solution:

- Development of a 3D efficient numerical method based on the coupling of the Semi-Analytical Method (SAM) and the eXtended-Finite Element Method (X-FEM):
  - SAM: Resolution of the 3D contact problem,
  - X-FEM: Simulation of the 3D crack propagation with frictional contact between the crack faces.



Example of a 3D simple case studied by Kaneta [KAN 86] with the body force method

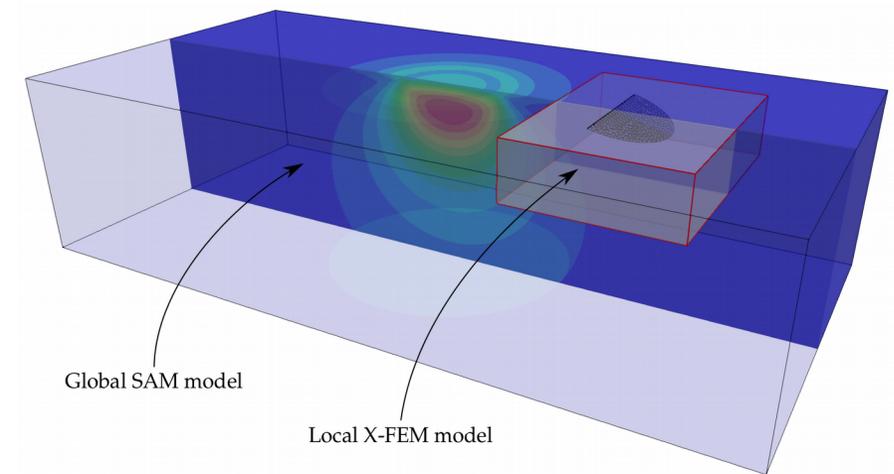


Illustration of the proposed SAM/X-FEM approach

[KAN 86] Kaneta M., Suetsugu M., Murakami Y. *Mechanism of Surface Crack Growth in Lubricated Rolling/Sliding Spherical Contact*. ASME. J. Appl. Mech, vol. 53 (2), 1986, p. 354–360.

# Outline



- **The SAM / X-FEM coupling approach**
- **Validation**
- **A Practical Example**
- **Conclusion and Perspectives**

# Outline



- **The SAM / X-FEM coupling approach**
- Validation
- A Practical Example
- Conclusion and Perspectives

# The SAM / X-FEM coupling approach

## Semi Analytical Method (SAM)

- Method dedicated to the resolution of 3D complex contact problem and initially proposed by Bentall and Johnson [BEN 67] and Paul and Hashemi [PAU 81],
- Based on the superposition of elementary analytical solutions (Green's functions),
- Used of Conjugate Gradient Method (CGM) and FFT techniques [POL 99] to solving the set of equations/inequations which define the BCs at the contact interface,
- Main assumptions:
  - Half-space bodies,
  - Small strains and small displacements,
  - Regular mesh density along each direction.

[BEN 67] Bentall R. H., Johnson K. L. *Slip in the Rolling Contact of Two Dissimilar Elastic Rollers*. International Journal of Mechanical Science, vol. 9 (6), 1967, p. 389-404.

[PAU 81] Paul B., Hashemi J. *Contact Pressure on Closely Conforming Elastic Bodies*. ASME Journal of Applied Mechanics, vol. 48 (3), 1981, p. 543-548.

[POL 99] Polonsky I. A., Keer L. M. *A Numerical Method for Solving Rough Contact Problems Based on the Multi-Level Multi-Summation and Conjugate Gradient Techniques*. Wear, vol. 231 (2), 1999, p. 206-219.

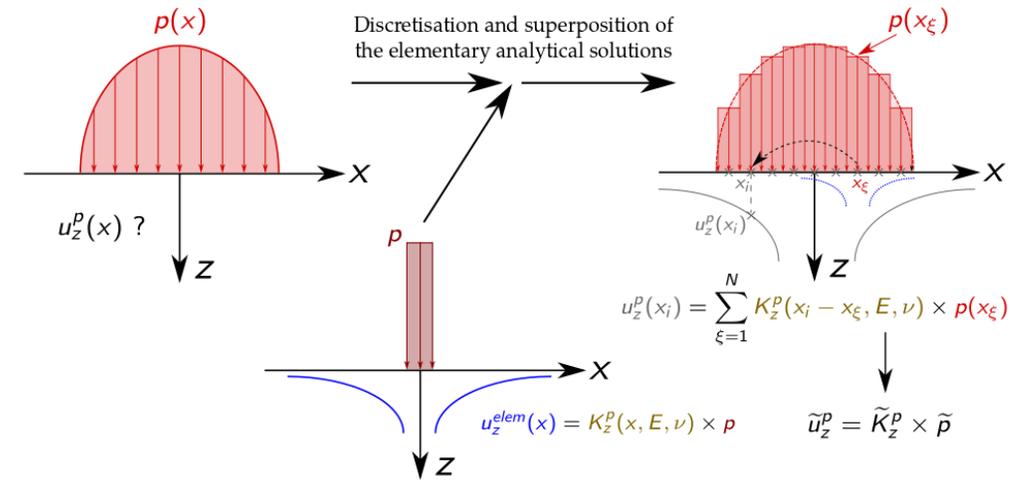
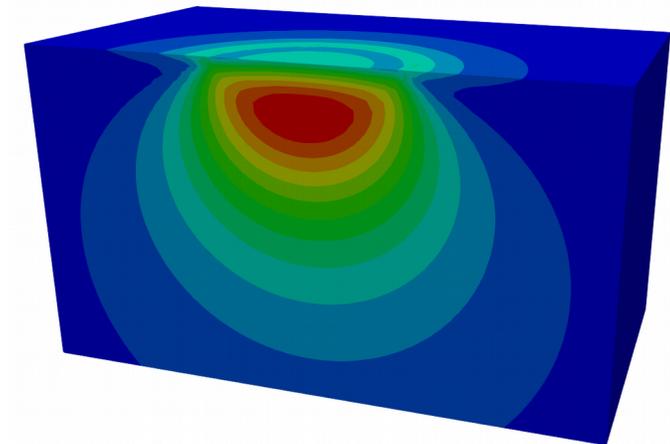


Illustration of SAM principle



Subsurface Von Mises Stresses extracted from SAM calculation

# The SAM / X-FEM coupling approach

## Semi Analytical Method (SAM)

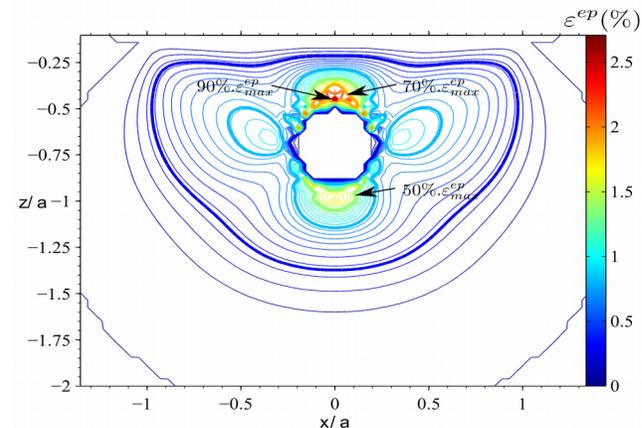
- **Main advantages:**

- Shorter CPU time compared to FEM,
- Better accuracy for a given mesh size,
- Easier pre-processing.

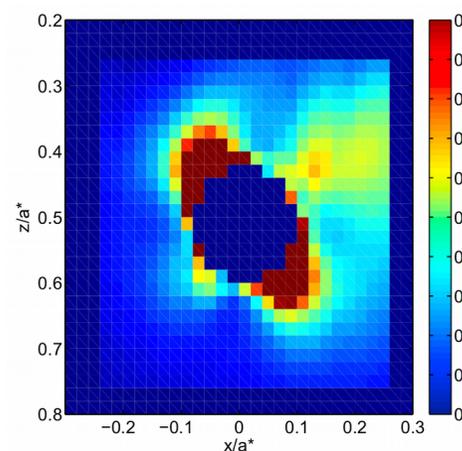
- **Implemented in a code named ISAAC<sup>®</sup>** and developed by the group of Prof D. Nélias at the LaMCoS laboratory,

- **Capabilities of ISAAC code:**

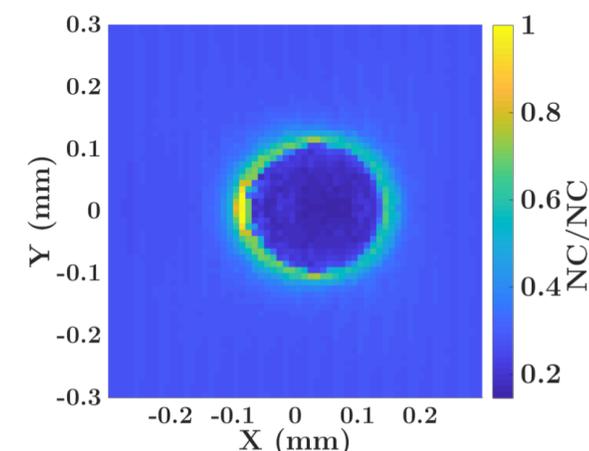
- Resolution of normal and tangential contact,
- Computation of subsurface stresses,
- Modeling of elastoplastic/viscoelastic behavior,
- Taking into account heterogeneity/inclusion,
- Calculation of surface wear,
- Modeling of material properties gradient, of material anisotropy, of surface coating.



Plastic strain distribution around stiff heterogeneity [AMU 16]



Modeling of butterfly wing formation around a stiff inclusion [BEY 19]



Simulation of crack nucleation around a dent [BON 20]

[AMU 16] Amuzuga K. V., Chaise T., Duval A., Nélias D. *Fully Coupled Resolution of Heterogeneous Elastic-Plastic Contact Problem*. ASME. J. Tribol, 2016, vol. 138 (2): 021403.

[BEY 19] Beyer T., Sadeghi F., Chaise T., Leroux J., Nélias D. *A coupled damage model and a semi-analytical contact solver to simulate butterfly wing formation around nonmetallic inclusions*. International Journal of Fatigue, vol. 127, 2019, p. 445-460.

[BON 20] Bonetto A. *Étude de l'indentation et de la fatigue des contacts roulants*. PhD thesis, INSA de Lyon. 2020.

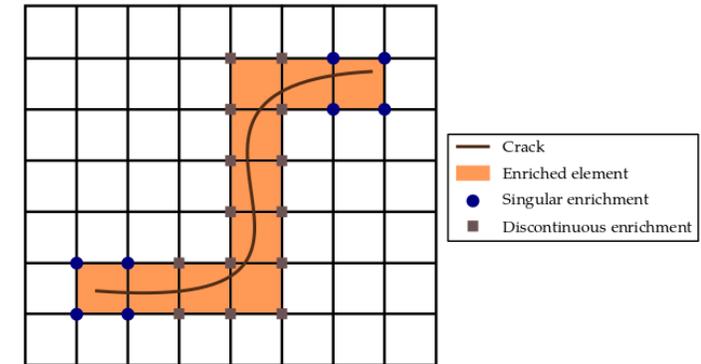
# The SAM / X-FEM coupling approach

## eXtended-Finite Element Method (X-FEM)

- Method dedicated to the 2D/3D crack growth simulation and initially introduced by Moës et al. [MOË 99],
- Based on the addition of local enrichment functions into the classical FE displacement approximation (local partition of unity):

$$\underline{u}(x) = \sum_{i \in N} N_i(x) u_i + \sum_{i \in N^e} N_i(x) \Phi(x) u_i^e$$

- Usually coupled with the level set technique to handle arbitrary 3D non planar cracks [MOË 02],
- Main advantages:
  - Crack / X-FEM mesh decoupling (easier pre-processing),
  - Better accuracy than FEM for a given mesh size,
  - Re-meshing not required during crack propagation.



Enrichment strategy of X-FEM

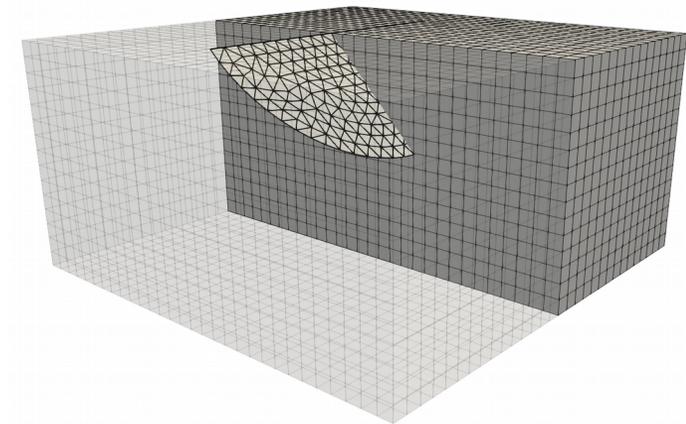


Illustration of X-FEM mesh independence

[MOË 99] Moës N., Dolbow J., Belytschko T. *A finite element method for crack growth without remeshing*. International Journal for Numerical Methods in Engineering, vol. 46, 1999, p. 131–150.

[MOË 02] Moës N., Gravouil A., Belytschko T. *Non-planar 3D crack growth by the extended finite element and level sets - Part I : Mechanical model*. International Journal for Numerical Methods in Engineering, vol.53, 2002, p. 2549–2568.

# The SAM / X-FEM coupling approach

## 3D X-FEM frictional contact crack model

- Model based on a two scales strategy (structure / crack),
- Introduction of a three-field weak formulation:

$$\begin{aligned}
 0 = & - \int_{\Omega} \sigma : \varepsilon(u^*) d\Omega + \int_{\Omega} f \cdot u^* d\Omega + \int_{\partial_2 \Omega} F \cdot u^* dS + \int_{\Gamma} \lambda \cdot u^* dS \\
 & + \int_{\Gamma} (t - \lambda) \cdot w^* dS \\
 & + \int_{\Gamma} (u - w) \cdot \lambda^* dS \quad \text{Weak coupling between } u \text{ et } w
 \end{aligned}$$

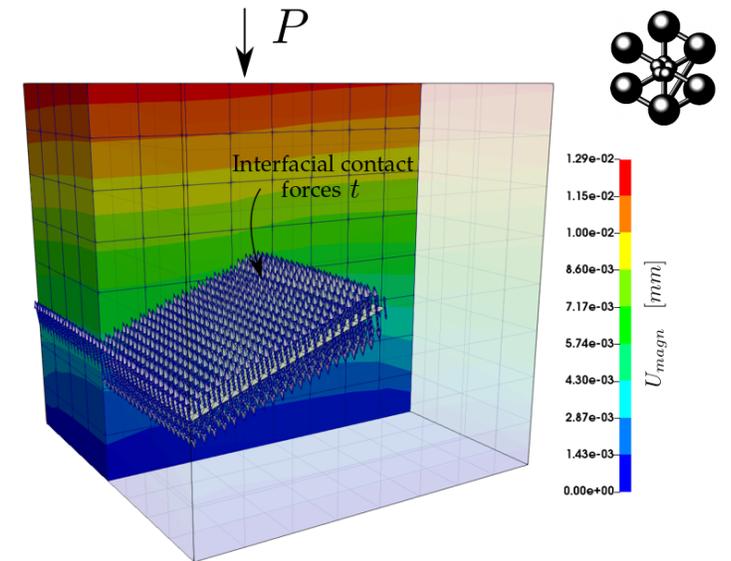
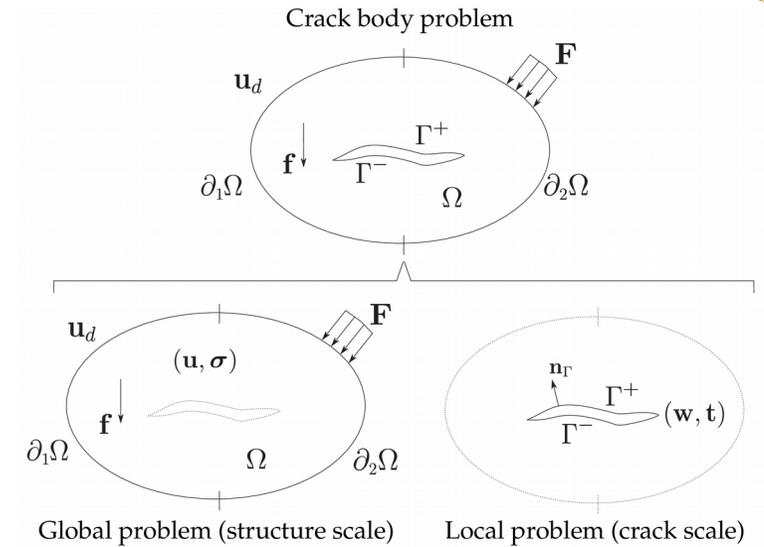
+ constitutive law in bulk  
+ frictional contact law at the crack interface

- Discretization of the three-field weak formulation within a X-FEM framework and resolution with the incremental LATIN method,
- Method developed by the group of Prof. A. Gravouil at LaMCoS [PIE 10, GRA 11] and implemented in the Cast3M code by B. Trollé [TRO 14].

[PIE 10] Pierres E., Baietto M.-C., Gravouil A. *A two-scale extended finite element method for modeling 3D crack growth with interfacial contact*. Computer Methods in Applied Mechanics and Engineering, vol. 199, 2010, p. 1165-1177.

[GRA 11] Gravouil A., Pierres E., Baietto M.-C. *Stabilized global-local X-FEM for 3D non-planar frictional crack using relevant meshes*. Int. J. Numer. Meth. Engng., vol. 88, 2011, p. 1449-1475.

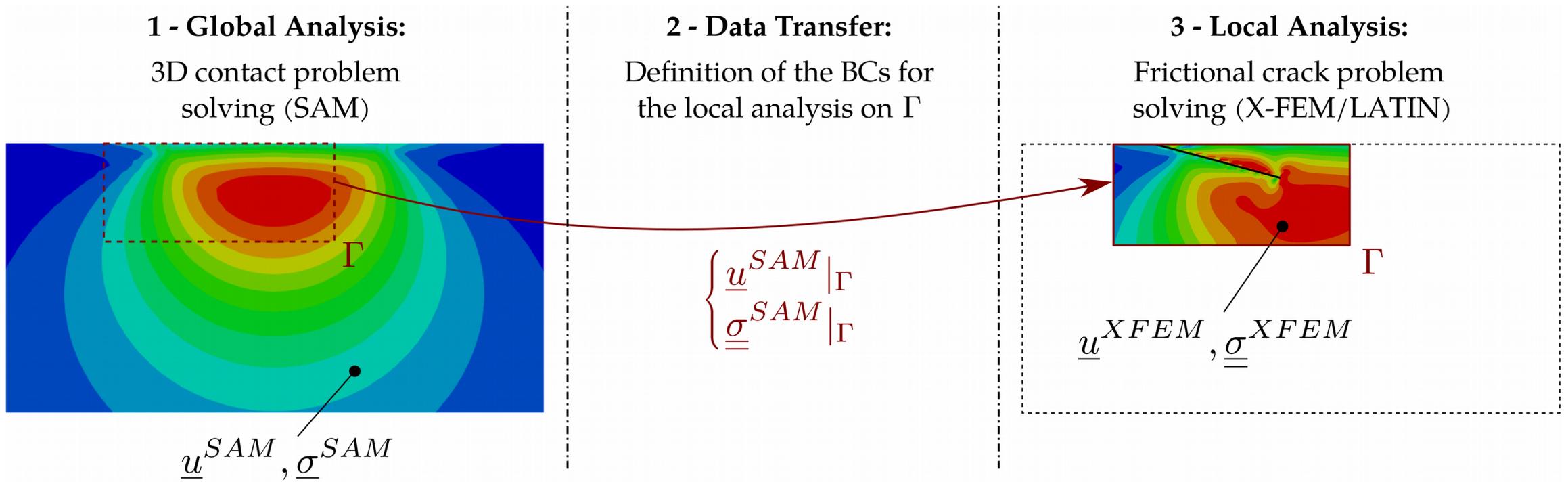
[TRO 14] Trollé B. *Simulation multi-échelles de la propagation des fissures de fatigue dans les rails*. PhD thesis, INSA de Lyon, 2014.



# The SAM / X-FEM coupling approach

## Main idea

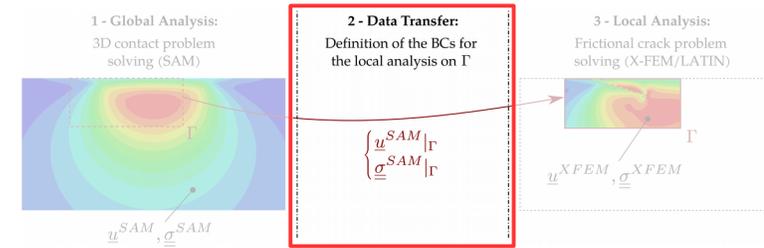
- **Hypothesis:**
  - Frictional crack problem has no significant influence on the contact problem.
- **Development of a two scales SAM / X-FEM strategy based on a submodeling technique [COR 99] (or top-down approach):**



[COR 99] Cormier N. G., Smallwood B. S., Sinclair G. B., Meda G. *Aggressive submodelling of stress concentrations*. Int J Numer Methods Eng, vol. 46 (6), 1999, p. 889-909.

# The SAM / X-FEM coupling approach

## Data Transfer



- 2 quantities can be transferred from global model to local model surface boundary  $\Gamma$ :
  - Displacement field,
  - Stress/surface traction field.
- In the SAM / X-FEM framework, only the stress field is transferred because cracks cause a stiffness change,
- Procedure to define the force boundary conditions for the local analysis:

- 1) Linear interpolation of global model stresses on  $\Gamma_e$ :

$$\sigma_{ij}^{\Gamma_e}(\underline{x}) = [N(\underline{x})] \sigma_{ij}^{SAM}$$

- 2) Calculation of surface traction vector on  $\Gamma_e$ :

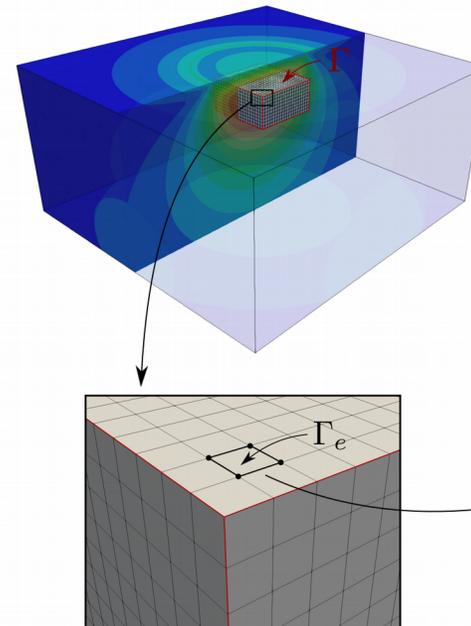
$$\underline{t}^{\Gamma_e}(\underline{x}) = \underline{\sigma}^{\Gamma_e}(\underline{x}) \cdot \underline{n}^{\Gamma_e}(\underline{x})$$

- 3) Integration of surface traction vector on  $\Gamma_e$ :

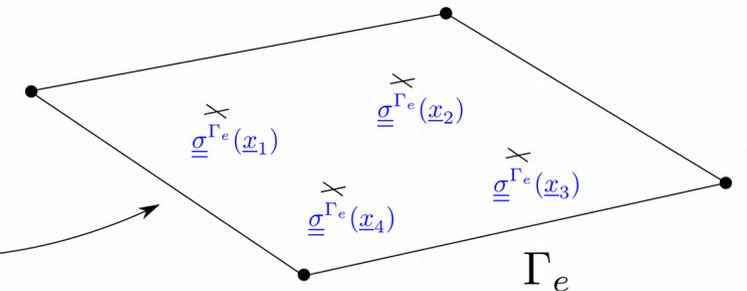
$$\{F^{\Gamma_e}\} = \iint_{\Gamma_e} [N^{\Gamma_e}(\underline{x})]^T \{t^{\Gamma_e}(\underline{x})\} d\Gamma$$

- 4) Summation of each elementary nodal force vector:

$$\{F^{\Gamma}\} = \sum \{F^{\Gamma_e}\}$$

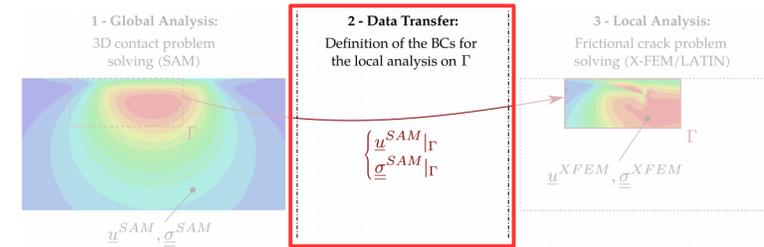


In blue : quantities defined at the Gauss points of  $\Gamma_e$   
 In Red : quantities defined at the nodes of  $\Gamma_e$



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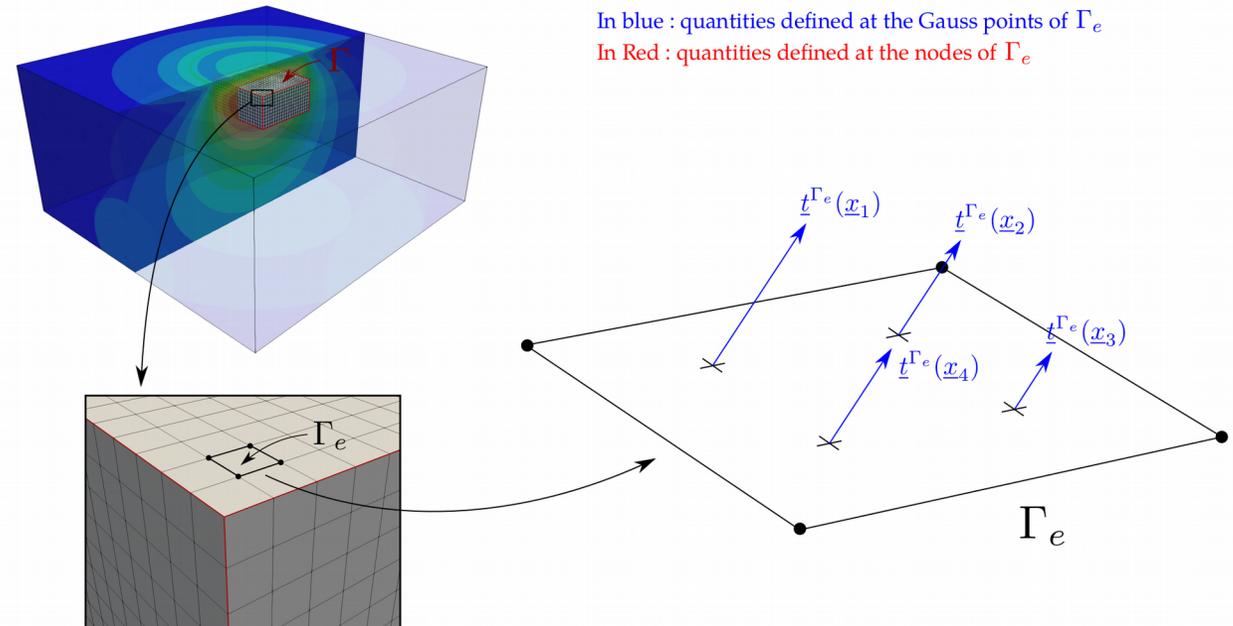
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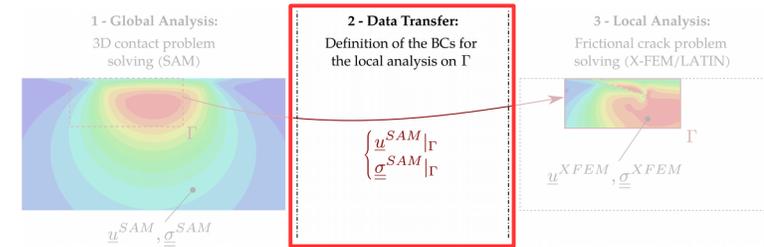
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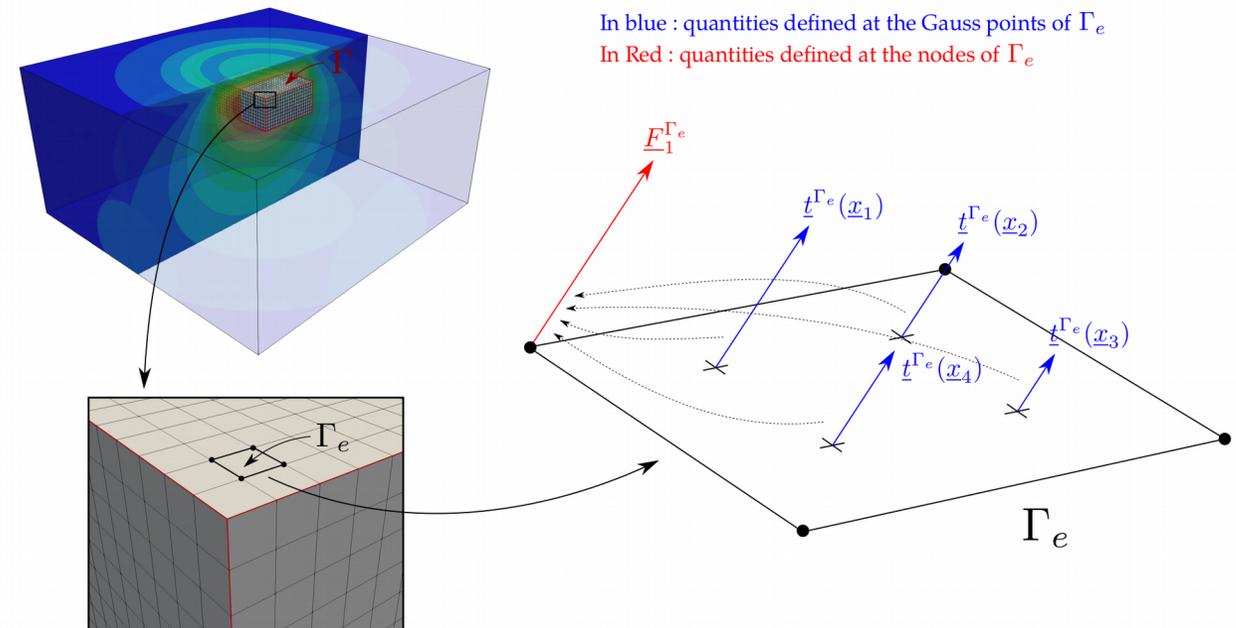
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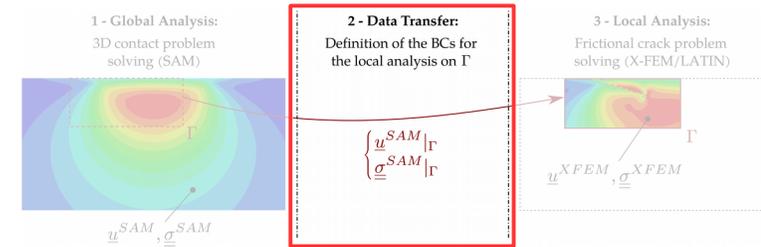
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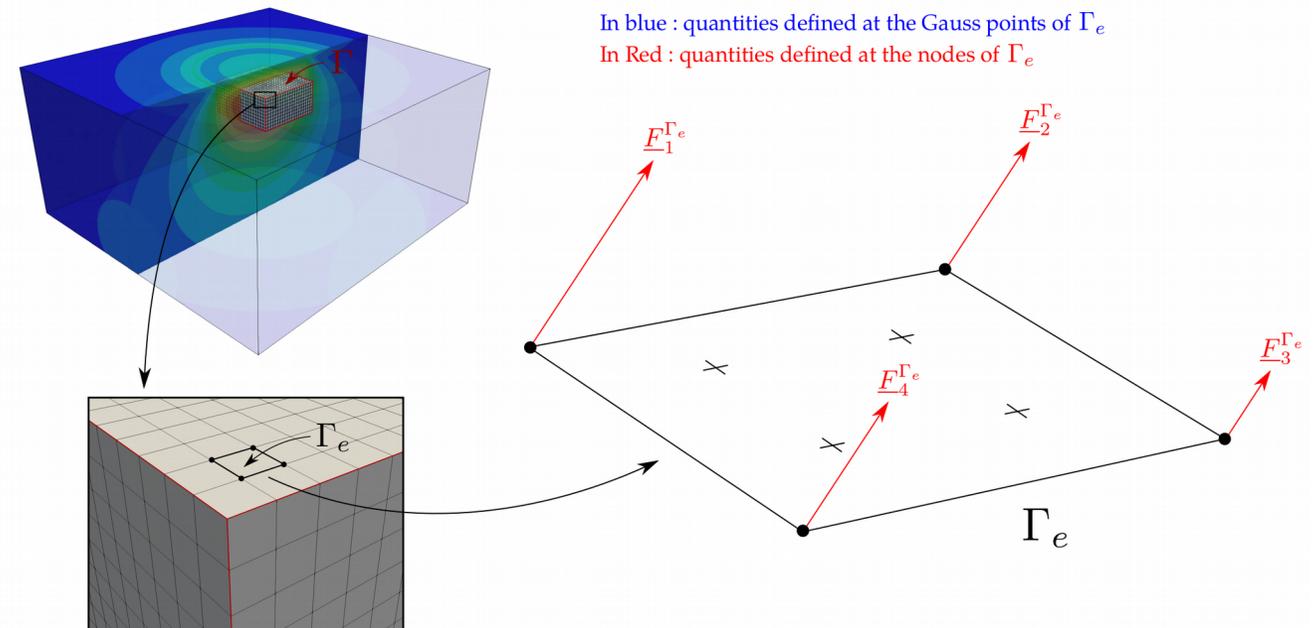
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# The SAM / X-FEM coupling approach

## Data Transfer

- Pure force boundary conditions exhibit rigid body motion issues caused by unbalanced nodal forces:

$$\sum \underline{F}^\Gamma \neq 0$$

- Root cause: interpolation step.

- Introduction of an operator  $P$  to balance the nodal force vector  $F^\Gamma$ :

$$F_p = PF^\Gamma = (I - R(R^T R)^{-1} R^T) F^\Gamma \implies R^T F_p = 0$$

- $P$  enforces the orthogonality condition between the balanced force vector  $F_p$  and the rigid body modes  $R$ .

- Initial self-balanced stresses (induced by surface hardening techniques) can be taken into account by adding this field to the global model stresses:

$$\underline{\underline{\sigma}}^{SAM} + \underline{\underline{\sigma}}^{initial}$$

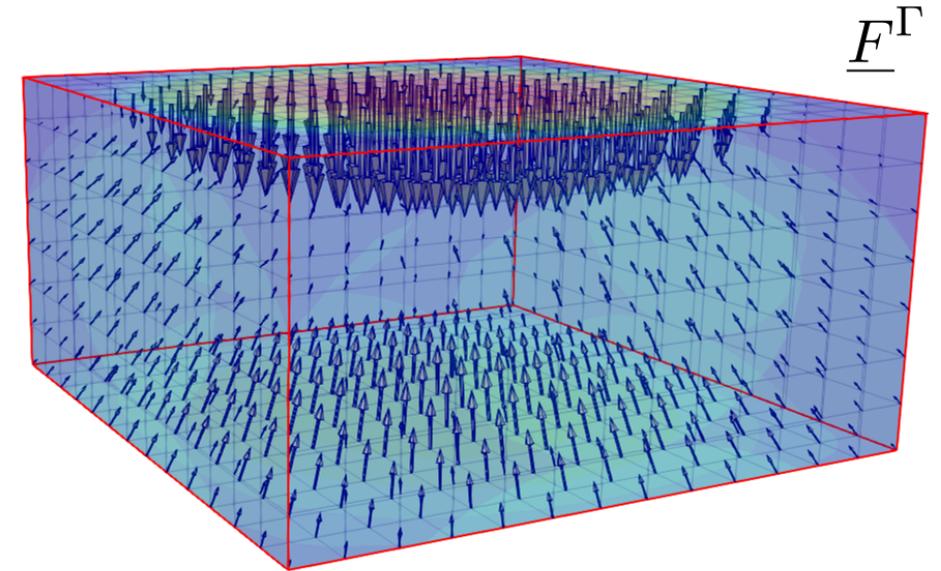
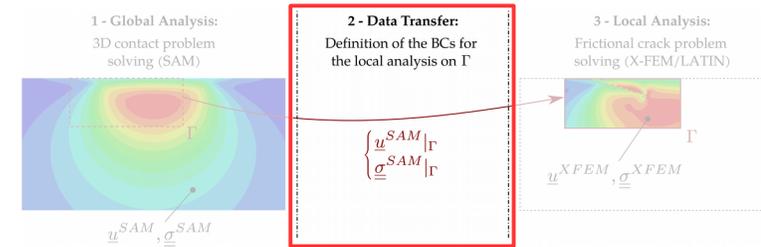
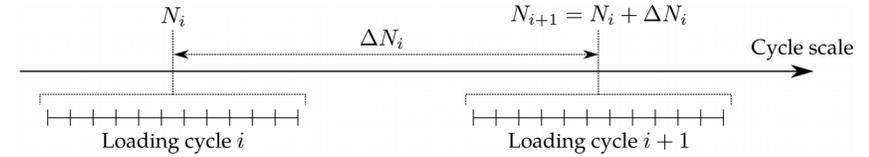


Illustration of nodal forces applied on local model surface boundary  $\Gamma$

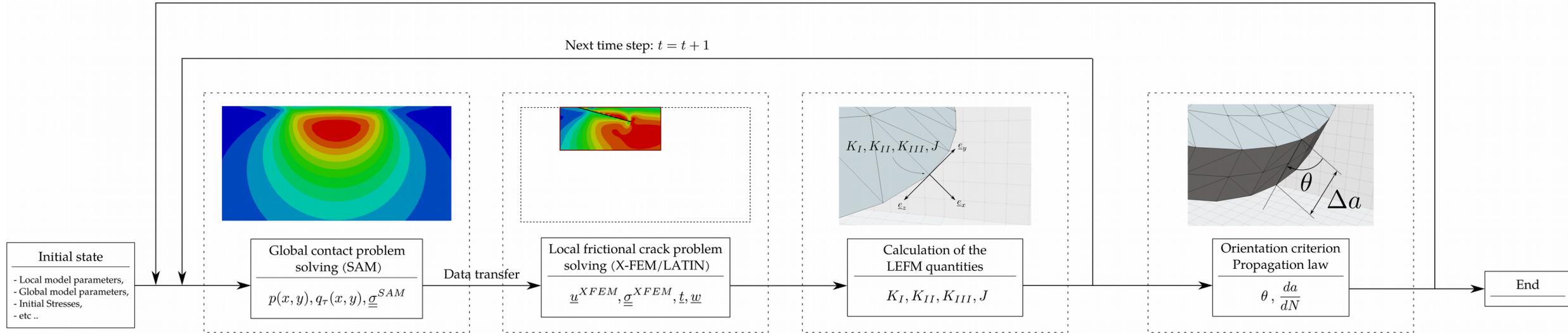
# The SAM / X-FEM coupling approach

## Application to crack propagation under rolling contact loading

- Algorithm for RCF crack growth simulation with the SAM / X-FEM method:



Next simulated cycle:  $N = N + \Delta N, a = a + \Delta a$



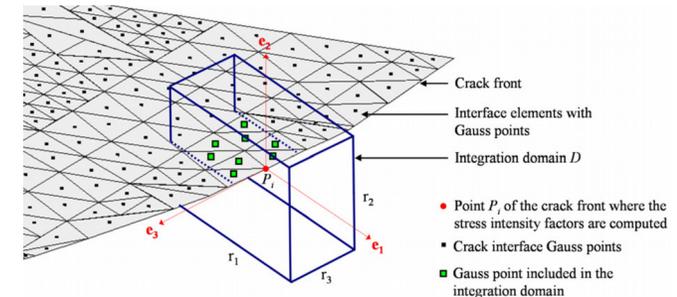
- Stress Intensity Factors (SIFs) calculation – 3D interaction integral [PIE 11, PRA 19]:

$$I_h = - \int_D (\sigma_{kl}^h \varepsilon_{kl}^{aux} \delta_{ij} - \sigma_{kj}^h u_{k,i}^{aux} - \sigma_{kj}^{aux} u_{k,i}^h) q_{i,j} dV - \int_{\Gamma_C^+ \cup \Gamma_C^-} t_k^h u_{k,1}^{aux} q_1 dS$$

Term accounting for contact and friction on crack faces

[PIE 11] Pierres E., Baietto M.-C., Gravouil A. *Experimental and numerical analysis of fretting crack formation based on 3D X-FEM frictional contact fatigue crack model*. Comptes Rendus Mécanique, Elsevier Masson, 2011, vol. 339 (7-8), p. 532-551.

[PRA 19] Prabel B. *Synthèse sur la méthode G-θ et son implémentation dans Cast3M*. Note Technique DEN, CEA, 2019.



# Outline



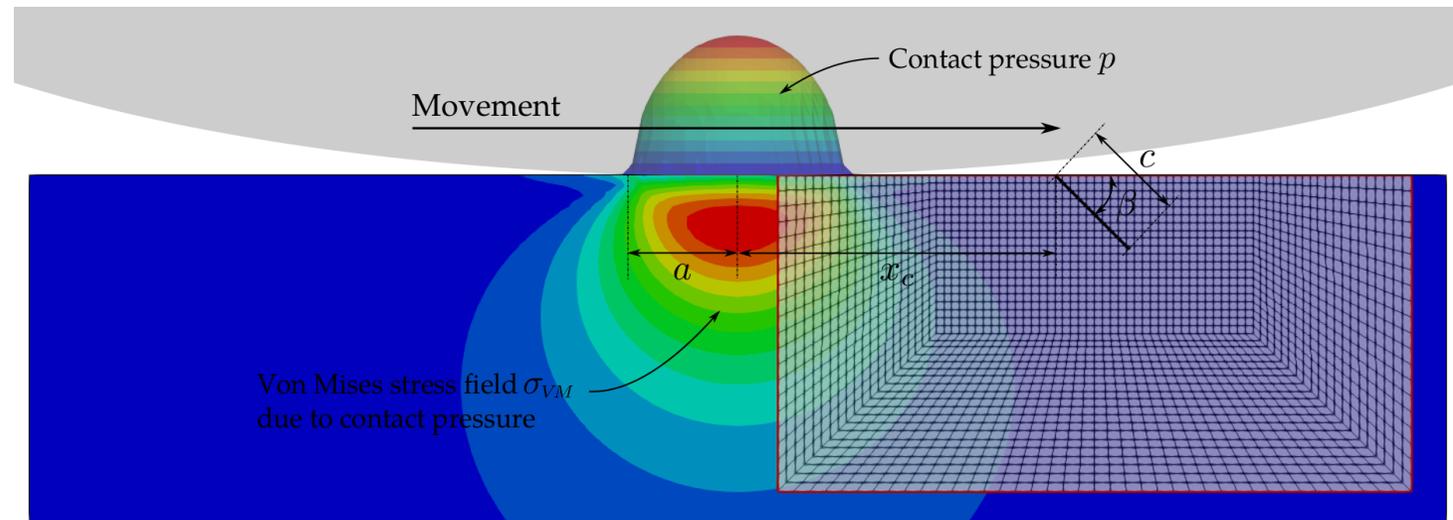
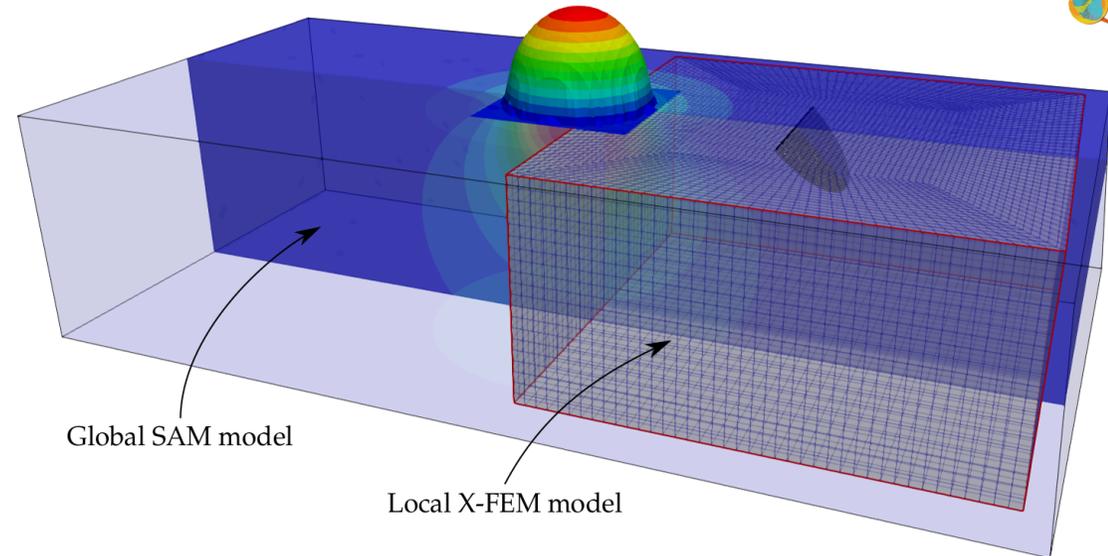
- The SAM / X-FEM coupling approach
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# Validation

## Comparison with Kaneta's results [KAN 86]

- **Model specifications:**

- Semi-circular crack under spherical contact,
- No contact between the crack faces,
- Crack angle:
  - $\beta = 45^\circ$
- Contact radius / Crack radius:
  - $a/c = 1.0$
- 2 contact friction coefficients:
  - $f = -0.1$
  - $f = 0.1$
- Local model dimensions:
  - $L_x = L_y = 6c$
  - $L_z = L_x / 2$



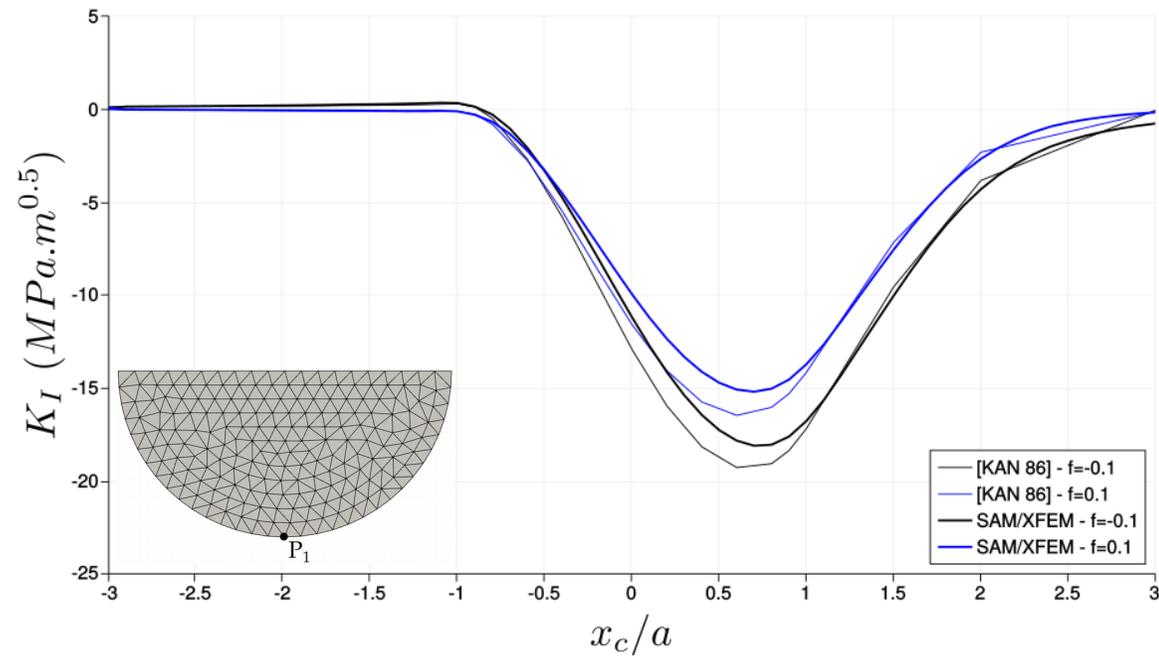
[KAN 86] Kaneta M., Suetsugu M., Murakami Y. *Mechanism of Surface Crack Growth in Lubricated Rolling/Sliding Spherical Contact*. ASME. J. Appl. Mech, vol. 53 (2), 1986, p. 354–360.

# Validation

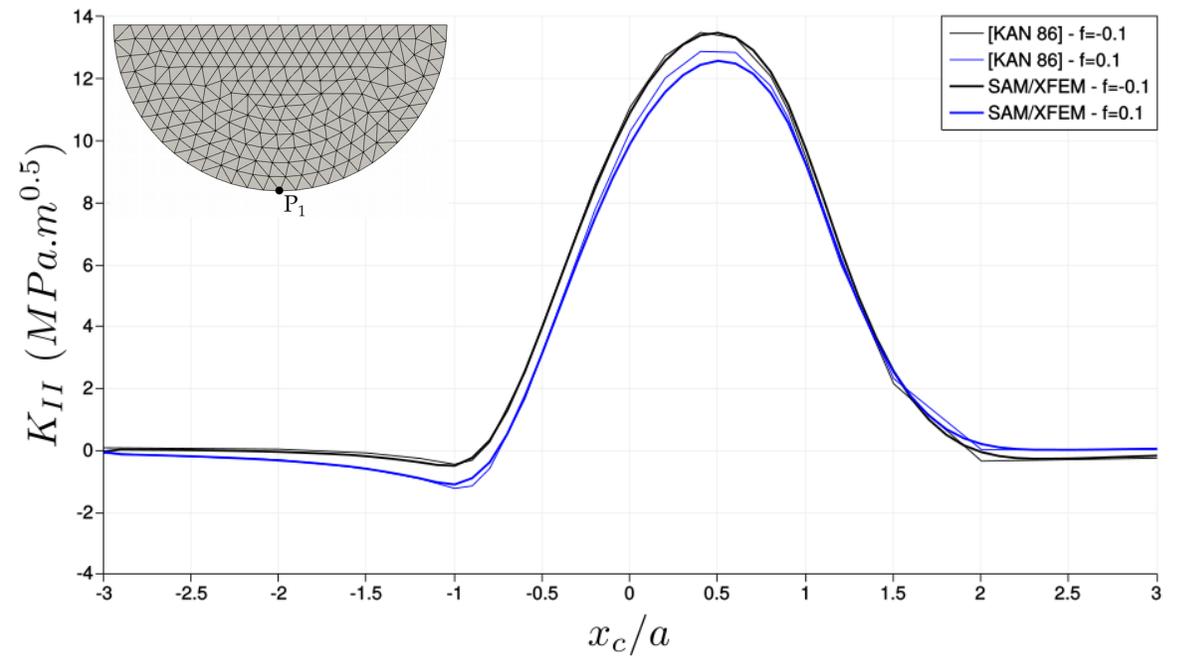
## Comparison with Kaneta's results [KAN 86]

- Results:

- SIFs variations at the  $P_1$  point –  $a/c = 1.0$



	f = -0.1		f = 0.1	
	$\Delta K_I$ (MPa.m <sup>0.5</sup> )	$\Delta K_{II}$ (MPa.m <sup>0.5</sup> )	$\Delta K_I$ (MPa.m <sup>0.5</sup> )	$\Delta K_{II}$ (MPa.m <sup>0.5</sup> )
SAM/X-FEM	18.45	13.97	15.25	13.67
[KAN 86]	19.57	13.93	16.45	14.10
Difference (%)	-5.72	+0.29	-7.29	-3.05



→ The results given by the developed SAM / X-FEM method show good agreement with the reference [KAN 86].

[KAN 86] Kaneta M., Suetsugu M., Murakami Y. *Mechanism of Surface Crack Growth in Lubricated Rolling/Sliding Spherical Contact*. ASME. J. Appl. Mech, vol. 53 (2), 1986, p. 354–360.

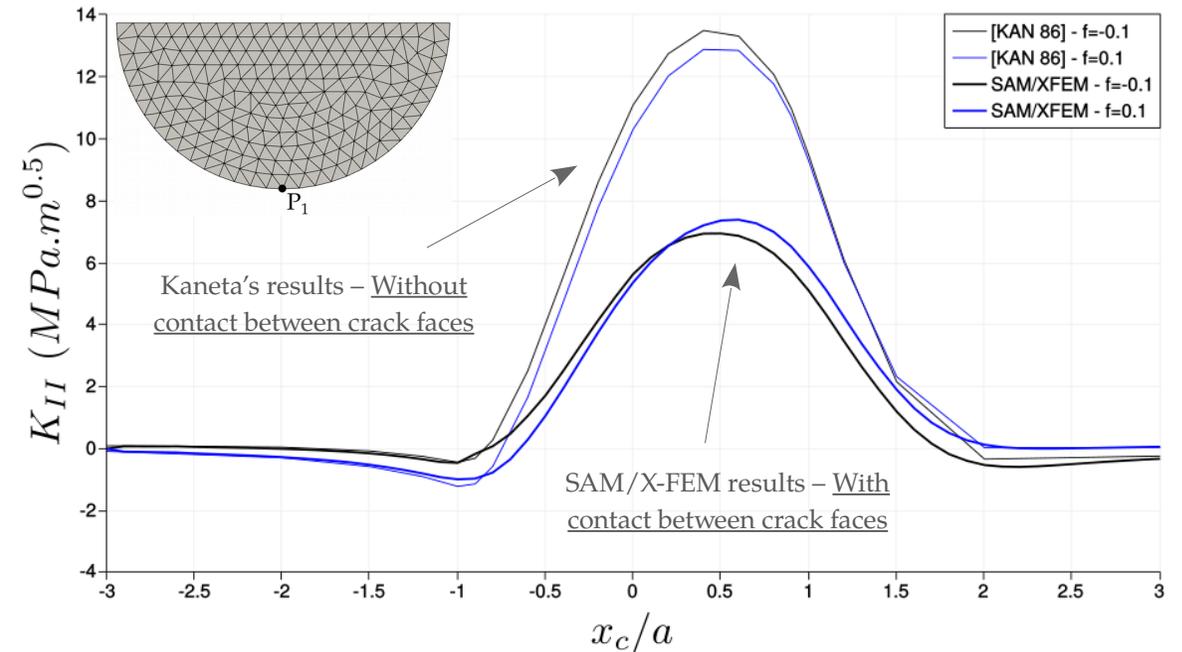
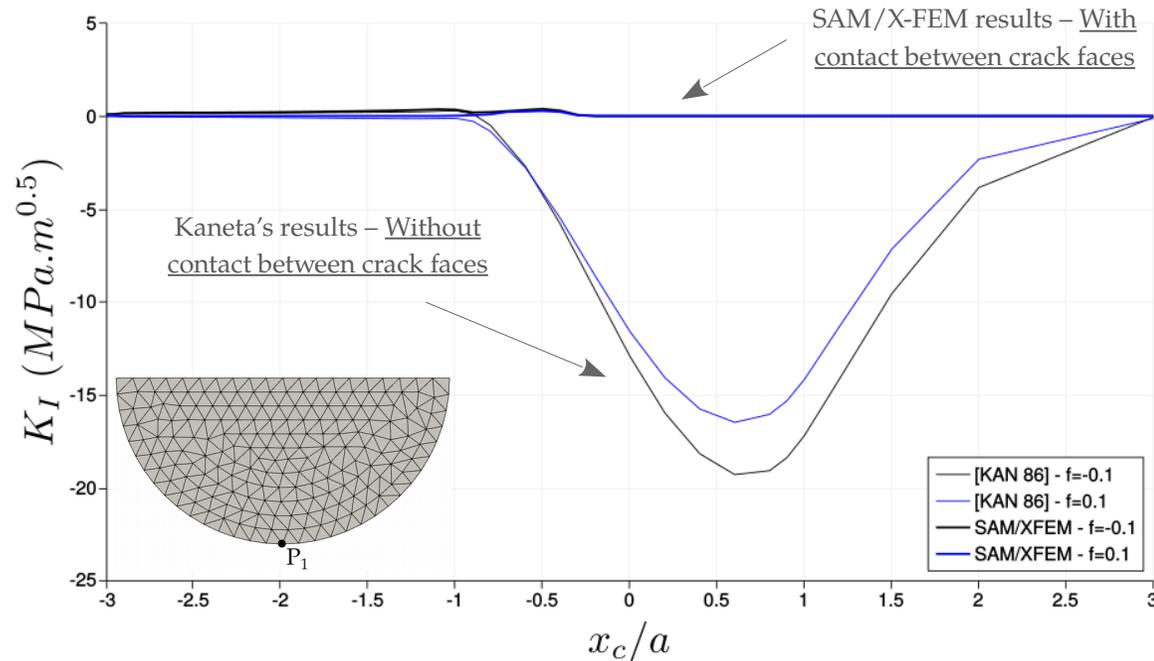
# Validation

## Comparison with Kaneta's results [KAN 86]

- Consideration of the interfacial contact between the crack faces:

- SIFs variations at the  $P_1$  point -  $a/c = 1.0 - f_c = 0$

	f = -0.1	f = 0.1
	$\Delta K_{II}$ (MPa.m <sup>0.5</sup> )	$\Delta K_{II}$ (MPa.m <sup>0.5</sup> )
SAM/X-FEM	7.53	8.38
[KAN 86]	13.93	14.10
Difference (%)	-45.94	-40,57



→ Consideration of the interfacial contact is essential for the simulation of crack propagation under rolling contact loading.

[KAN 86] Kaneta M., Suetsugu M., Murakami Y. *Mechanism of Surface Crack Growth in Lubricated Rolling/Sliding Spherical Contact*. ASME. J. Appl. Mech, vol. 53 (2), 1986, p. 354-360.

# Outline



- The SAM / X-FEM coupling approach
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- **A Practical Example**
- Conclusion and Perspectives

# A Practical Example

## Crack growth under rolling contact loading

- **Model specifications:**

- Semi-circular crack under spherical contact,
- Frictional contact between the crack faces,

- Contact parameters:

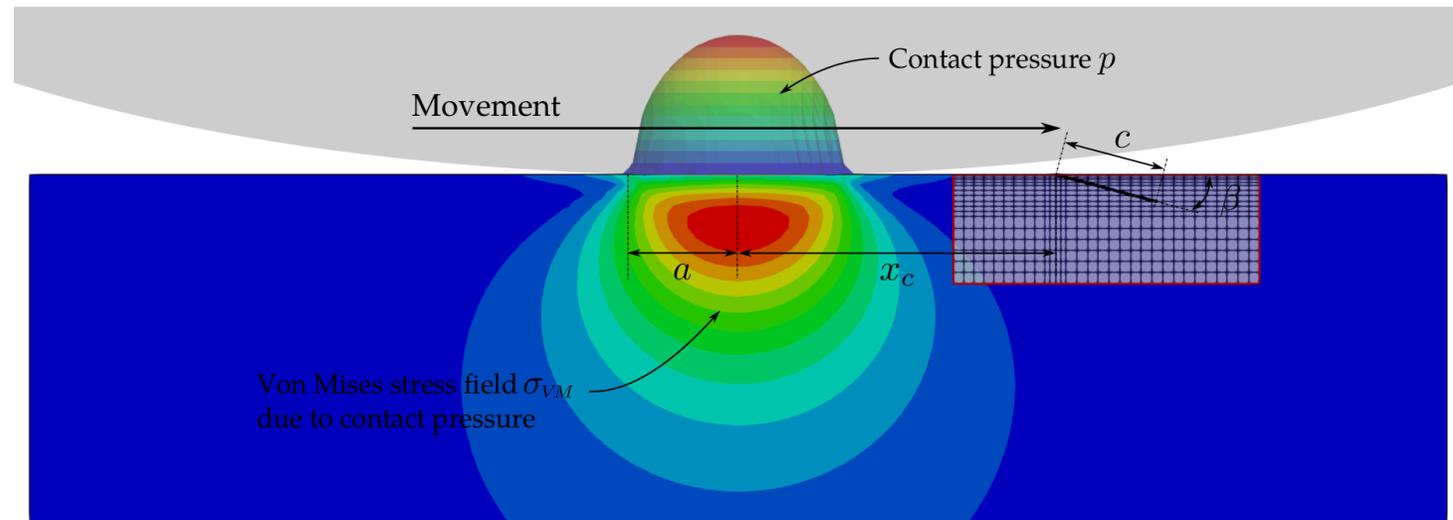
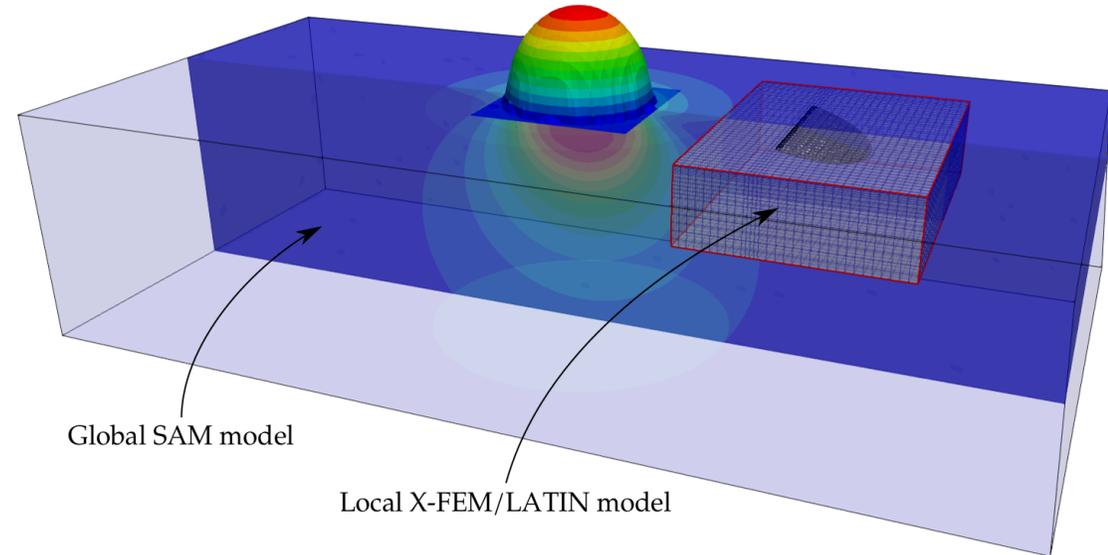
- $P_0 = 1468.5 \text{ Mpa}$ ,
- $a = 0.2 \text{ mm}$
- $f = 0$

- Crack parameters:

- $\beta = 15^\circ$
- $a/c = 1.0$
- $f_c = 0$

- Local model dimensions:

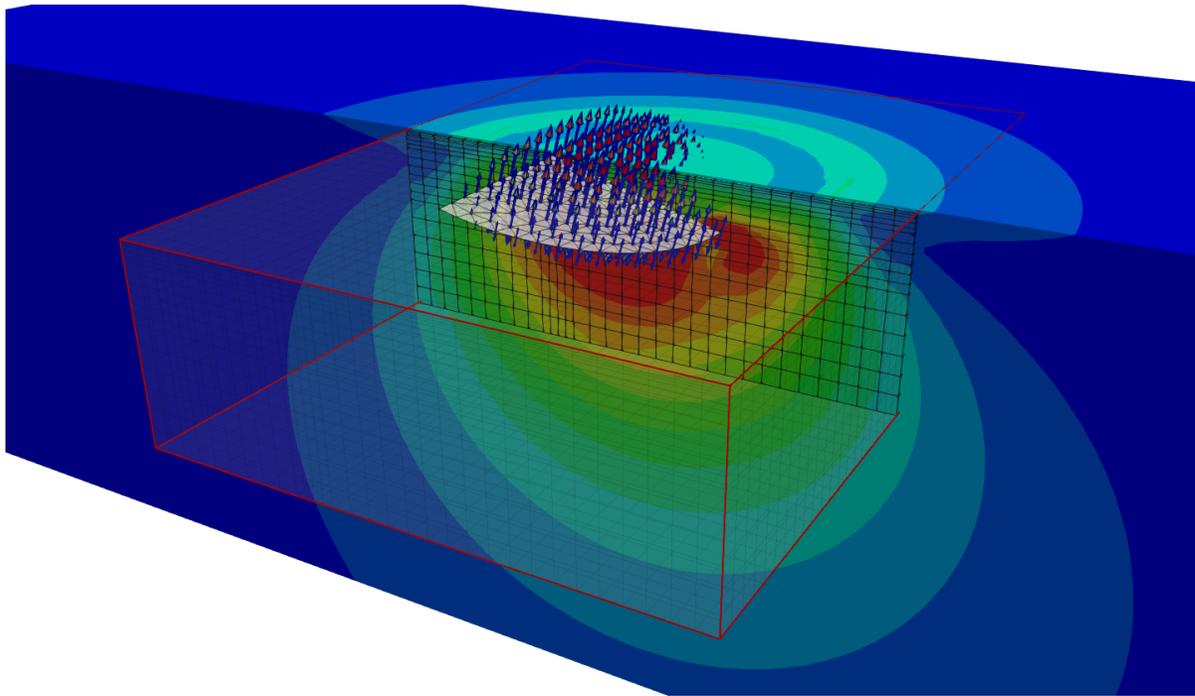
- $L_x = L_y = 2c$
- $L_z = 4c \cdot \sin(\beta)$



# A Practical Example

## Crack growth under rolling contact loading

- Results for the 1<sup>st</sup> loading cycle:

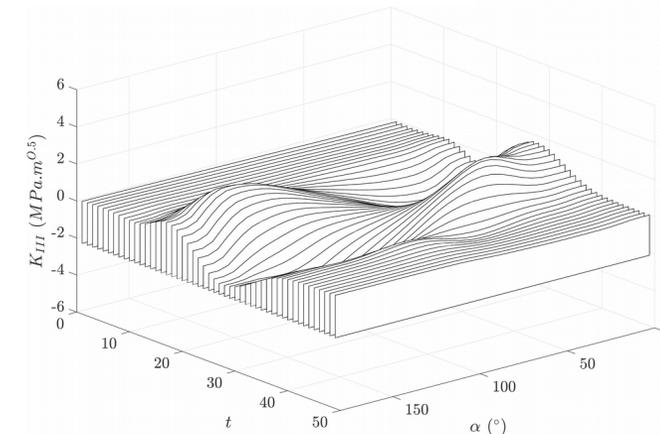
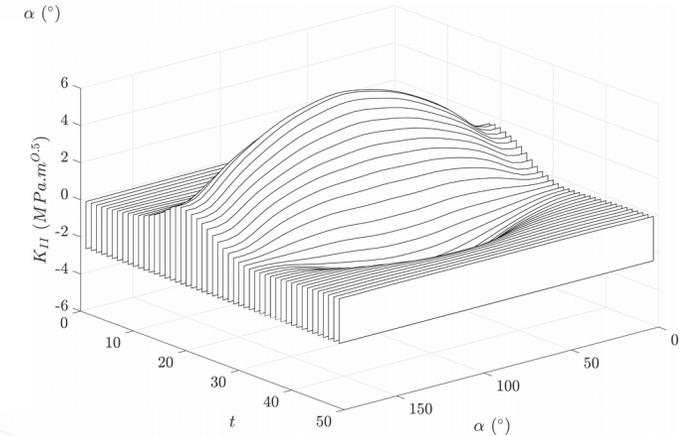
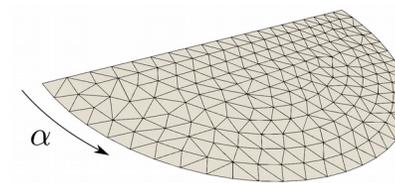
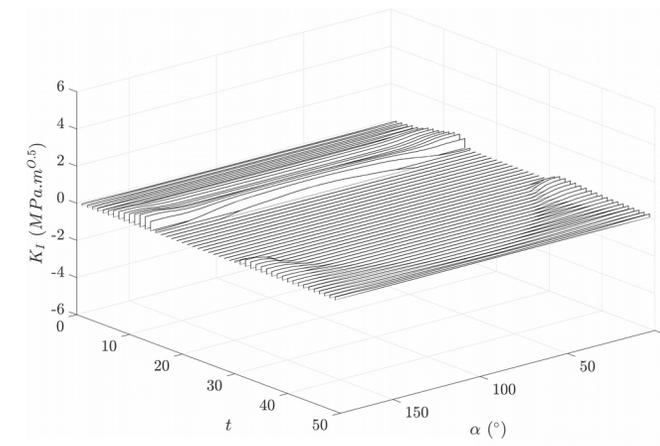


CPU time: 40 min on personal laptop (49 time steps)



→ RAM: 31Go

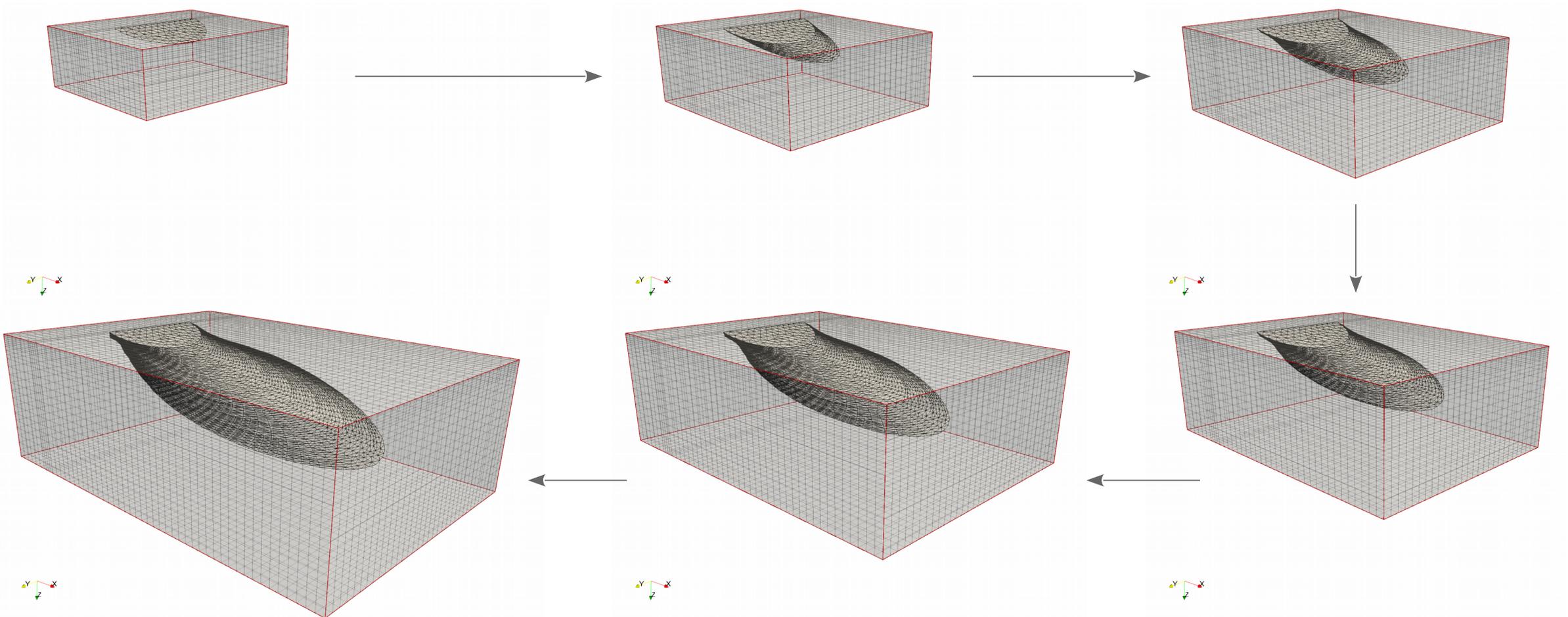
→ Processor: Intel® Core™ i7-7820HQ CPU @ 2.90GHz × 8



# A Practical Example

## Crack growth under rolling contact loading

- Mode II crack propagation over several cycle jumps:



# Outline

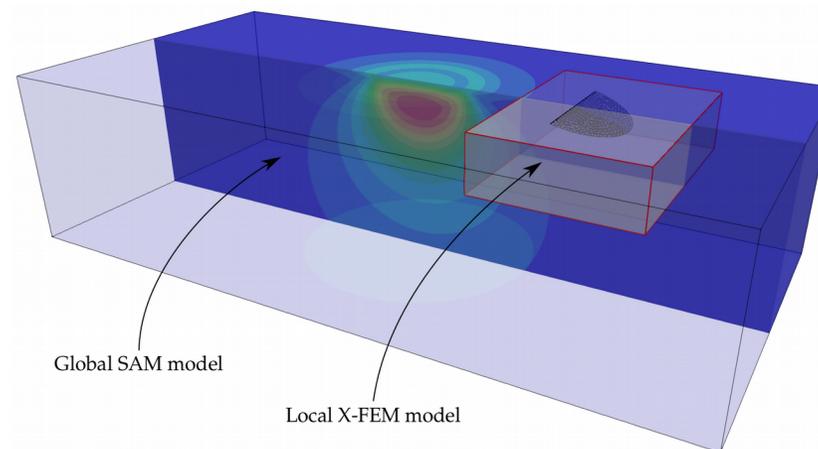


- The SAM / X-FEM coupling approach
- Validation
- A Practical Example
- Conclusion and Perspectives

# Conclusion & Perspectives

- **Key points:**

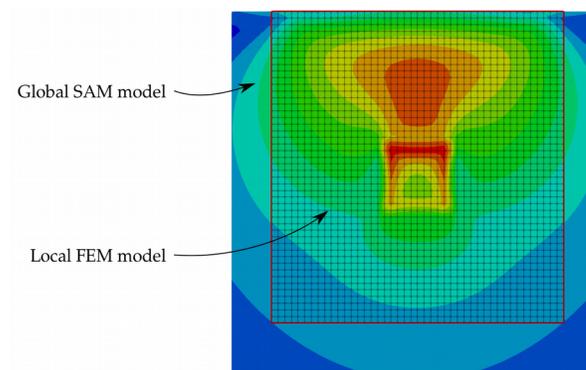
- Development of a novel method based on a SAM / X-FEM coupling,
- Capabilities of the developed approach:
  - 3D modeling of non-planar frictional crack propagation under contact loading,
  - Robust and accurate,
  - CPU time savings,
  - Easy pre-processing.
- Validation with results extracted from literature.



=> Great potential of the SAM / X-FEM method to understand the complex crack growth behavior under contact loading.

- **Other potential applications:**

- Crack propagation under fretting condition,
- Local plasticity,
- Heterogeneity problem,
- ...



*SAM / FEM coupling applied to heterogeneity problem*

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Thank you for your attention